



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Malleable Human Interfaces for Space Flight Applications

11 April 2002

LtCol Tom Duncavage

Mr. Joe Hamilton

Mr. John Jackson

NASA Johnson Space Center

Dr. Jim Whiteley, PE

Mr. Chris Slovacek

Mr. Chris Keller

Mr. Alan Gifford

Johnson Engineering



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

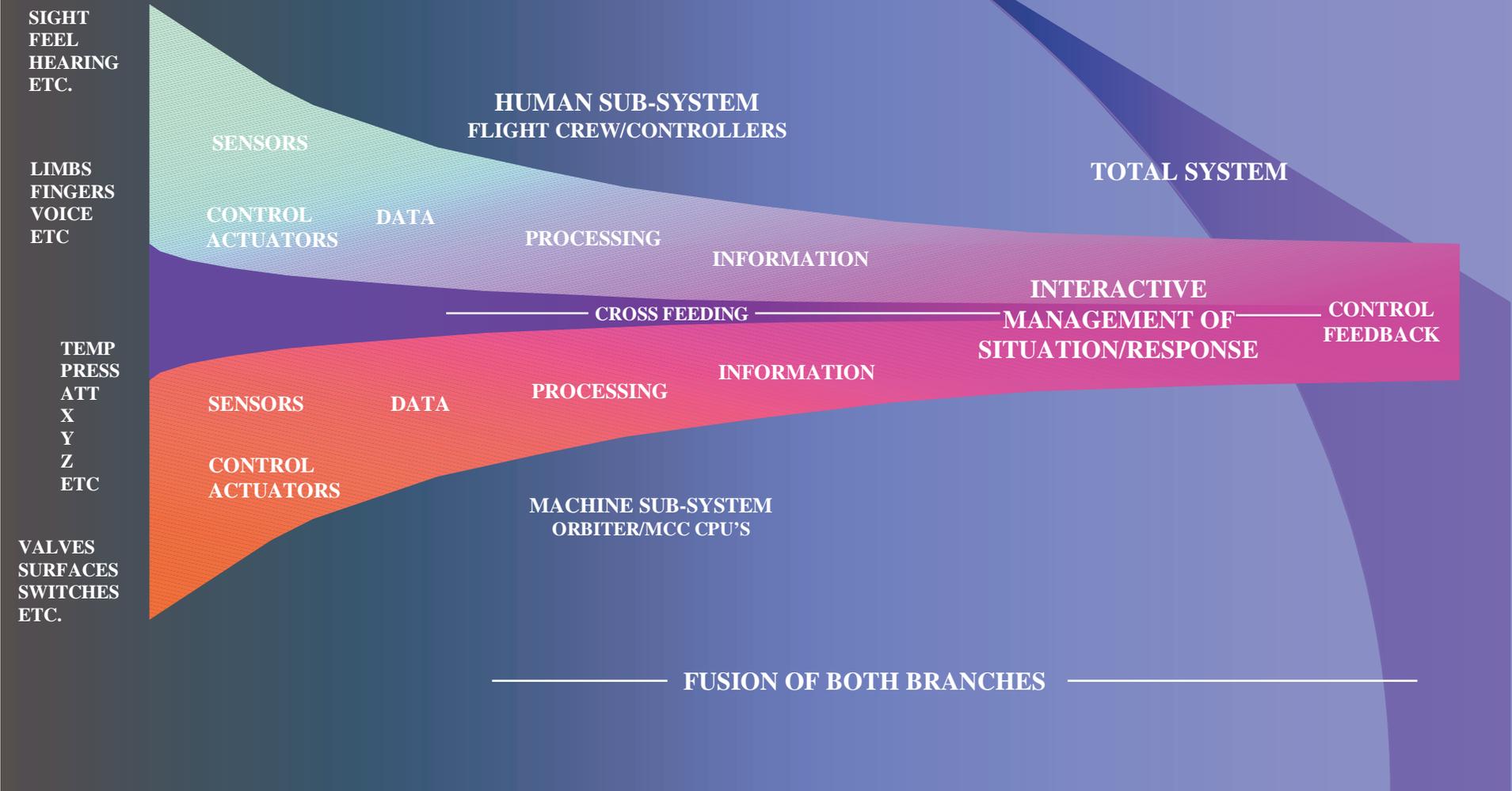
Malleable Human Interfaces (MHI) Purpose and Top-Level Goals

- The MHI project is an endeavor to develop conventions, guidelines and standards for malleable, transportable, consistent, and common crew interfaces for human space flight.
- MHI will demonstrate utility through performance measurement of MHI conceptual interfaces applied to specific spacecraft information systems.
- MHI envisioned products will have applicability for current operational and future long duration human space flight programs.



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

FUSION OF HUMAN AND MACHINE SENSORY DATA, PROCESSING, AND INFORMATION





**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

MHI Objectives

- Further develop and refine notional user interface designs
- Conduct performance measurement and testing
- Determine utility of notional malleable interface design(s)
- Develop and maintain interface and test result library
- Formulate MHI conventions, guidelines and standards from research results



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

MHI Old Timeline

- Oct 01 – Feb 02
 - Focused development of notional caution and warning interface designs
- March 02 – June 02
 - Conduct performance measurement and testing
 - Catalog results for analysis
- July 02 – Sept 02
 - Identify conventions, guidelines and standards relevant to caution and warning scenarios for human space flight
 - Document and present project findings as appropriate



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

MHI Timeline FY 02

- Oct 01 – Dec 01
 - Focused development of notional caution and warning interface designs
- Jan 02 – April 02
 - SLI interface design internal/external to vehicle
- May 02 – July 02
 - Conduct performance measurement and testing
 - Catalog results for analysis
- July 02 – Sept 02
 - Identify conventions, guidelines and standards relevant to SLI scenarios for human space flight
 - Document and present project findings as appropriate



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

MHI FY-2002 Milestones

- Complete MHI notional caution and warning interface(s)
- Complete performance measurement and data collection
- Analyze project research results
- Complete project summary report



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Work Performed – Internal Shuttle/SLI

- Updated the Dual Seat session in PRISMS to accommodate the forward section of the Shuttle's cockpit.
- Developed head/torso/hand model.
- Added the ability to turn virtual displays on/off using the stylus.



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Work Performed – External Shuttle/SLI

- Created a preliminary model in ProE for an SLI rocket with a CRV-like crew area at the top. Imported that model into a Prisms session to begin the evaluation of preliminary SLI concepts.
- Imported the position & orientation data for an ejection sequence of an escape pod. Created a flight path to illustrate the path in the CES (Crew Escape System) session.



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Work Performed – External Shuttle/SLI

- Converted and imported a database of positions and orientation of a typical flight of the shuttle. From these values, created a visual flight path and also attached a notional SLI vehicle to fly along the flight path with the new motion rules.
- Began insertion of display outputs for the independent variables that will be modified and monitored during the Crew Escape System studies in PRISMS.



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Work Performed – External Shuttle/SLI

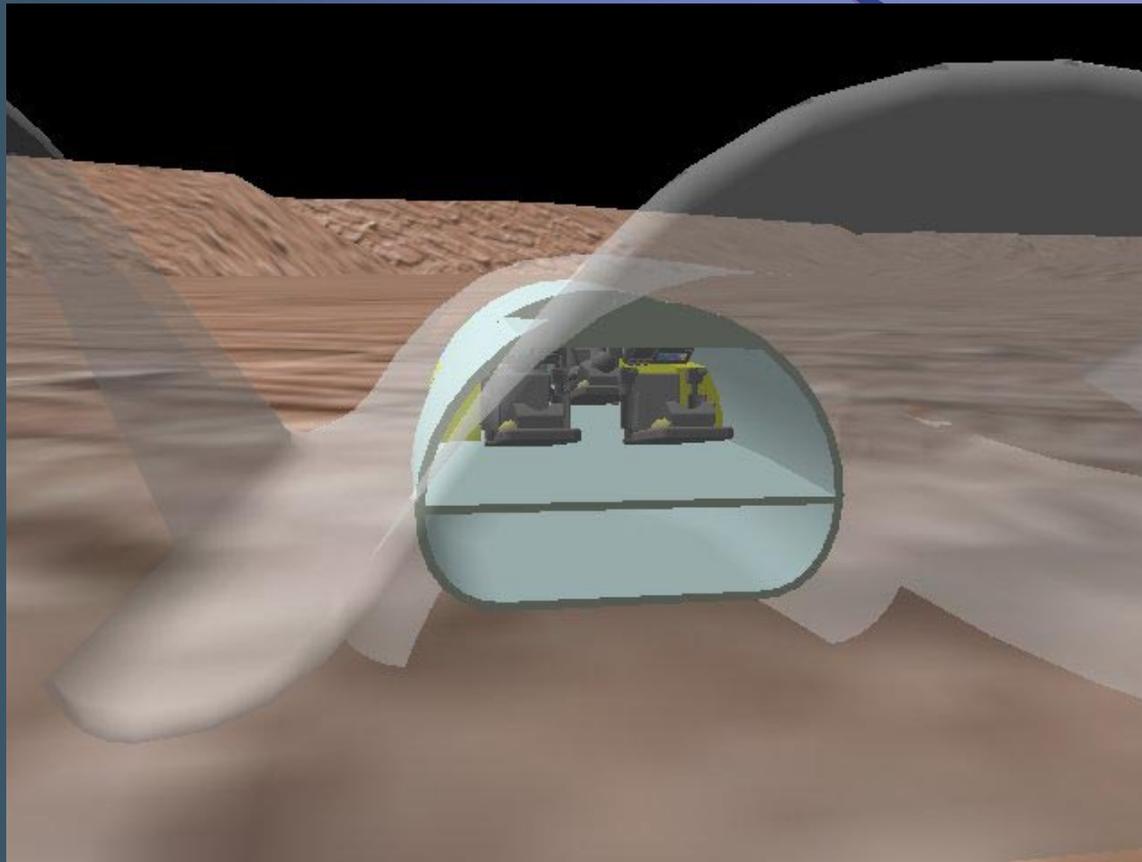
- Created an illustration of an explosion within the SLI session. This will be used in conjunction with the escape trajectory to determine the optimal sequence of events and visualization of appropriate crew/ground control annunciation.
- Variables such as blast model, time of explosion, time of escape system activation, and others can now be modified and monitored on an on-screen display by the experimenters.
- Created the rules and derivations to control the escape pod's activation and movement.



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

CRV/SLI CTV

Exterior





**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

CRV/SLI CTV

Interior





**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

CRV/SLI CTV

Common Interface Set





**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Shuttle Common Interface Set Translation

Inside the cockpit





**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Virtual Interactive Crewman



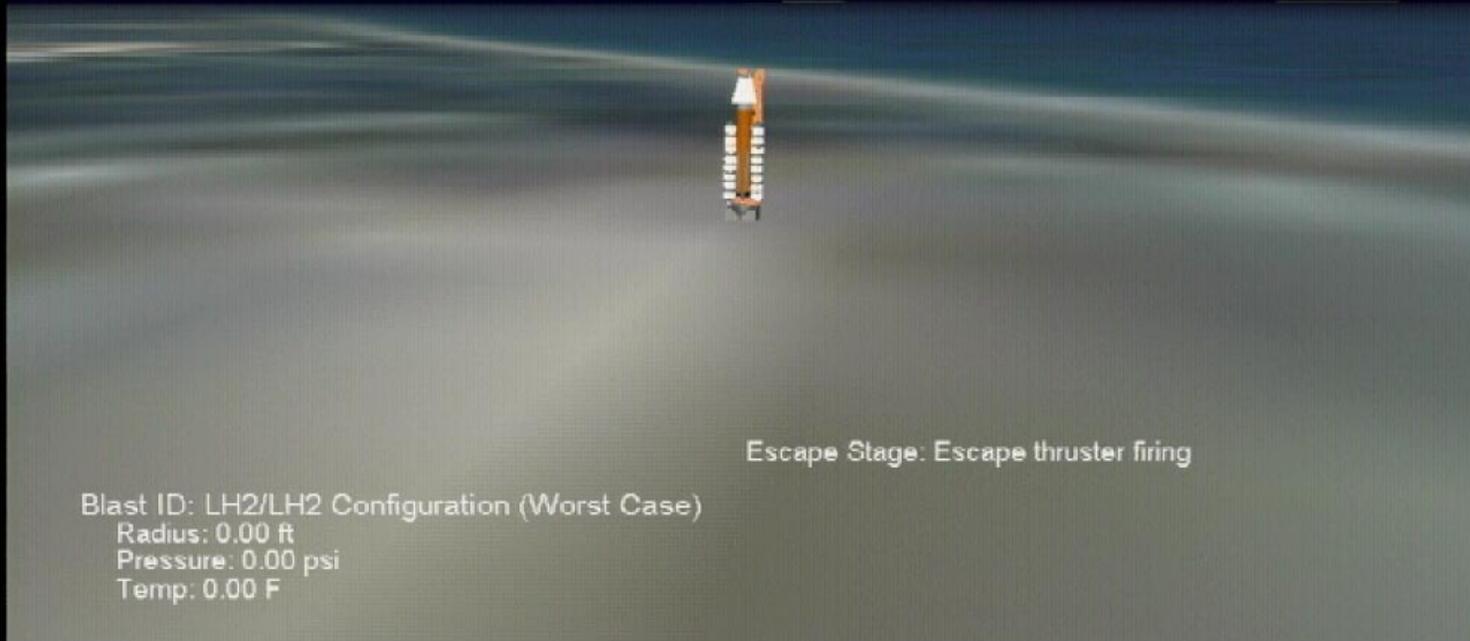


**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Catastrophic Failure Visualization Sequence

Time: 1.98 sec
SimSpeed: 10.00 %

Run ID: 0001
Take-off @ 0.00 sec
Escape activation @ 0.50 sec
Explosion @ 2.45 sec



Escape Stage: Escape thruster firing

Blast ID: LH2/LH2 Configuration (Worst Case)
Radius: 0.00 ft
Pressure: 0.00 psi
Temp: 0.00 F

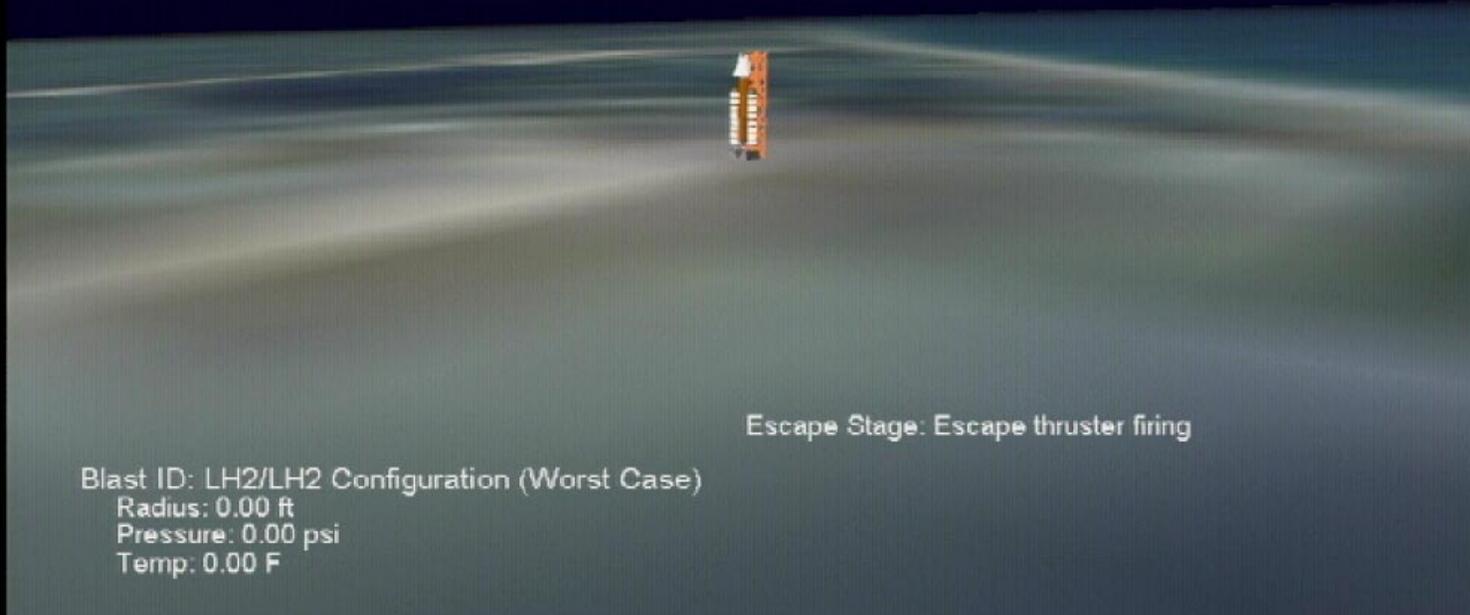


**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Catastrophic Failure Visualization Sequence

Time: 2.41 sec
SimSpeed: 10.00 %

Run ID: 0001
Take-off @ 0.00 sec
Escape activation @ 0.50 sec
Explosion @ 2.45 sec



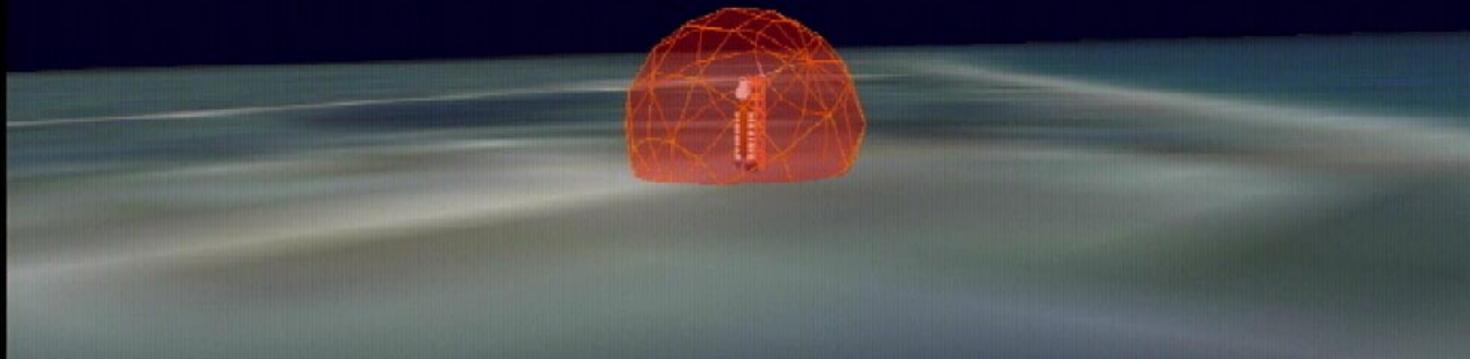


**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Catastrophic Failure Visualization Sequence

Time: 2.52 sec
SimSpeed: 10.00 %

Run ID: 0001
Take-off @ 0.00 sec
Escape activation @ 0.50 sec
Explosion @ 2.45 sec



Escape Stage: Escape thruster firing

Blast ID: LH2/LH2 Configuration (Worst Case)
Radius: 204.58 ft
Pressure: 67.84 psi
Temp: 0.00 F



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Catastrophic Failure Visualization Sequence

Time: 2.58 sec
SimSpeed: 10.00 %

Run ID: 0001
Take-off @ 0.00 sec
Escape activation @ 0.50 sec
Explosion @ 2.45 sec



Escape Stage: Escape thruster firing

Blast ID: LH2/LH2 Configuration (Worst Case)
Radius: 397.97 ft
Pressure: 48.17 psi
Temp: 0.00 F

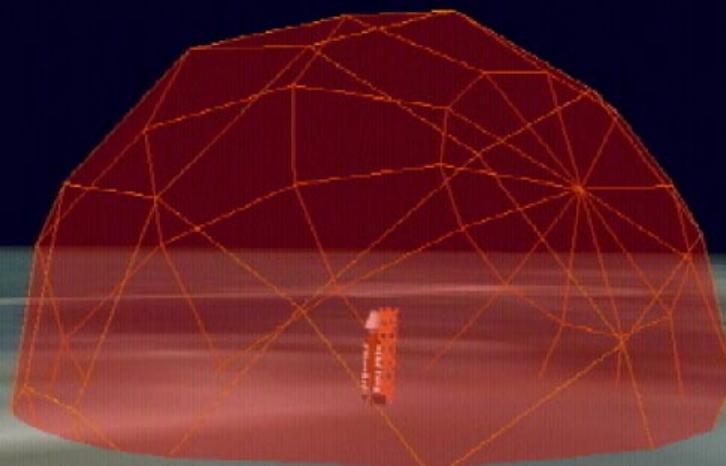


**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Catastrophic Failure Visualization Sequence

Time: 2.70 sec
SimSpeed: 10.00 %

Run ID: 0001
Take-off @ 0.00 sec
Escape activation @ 0.50 sec
Explosion @ 2.45 sec



Escape Stage: Escape thruster firing

Blast ID: LH2/LH2 Configuration (Worst Case)
Radius: 590.63 ft
Pressure: 20.32 psi
Temp: 0.00 F

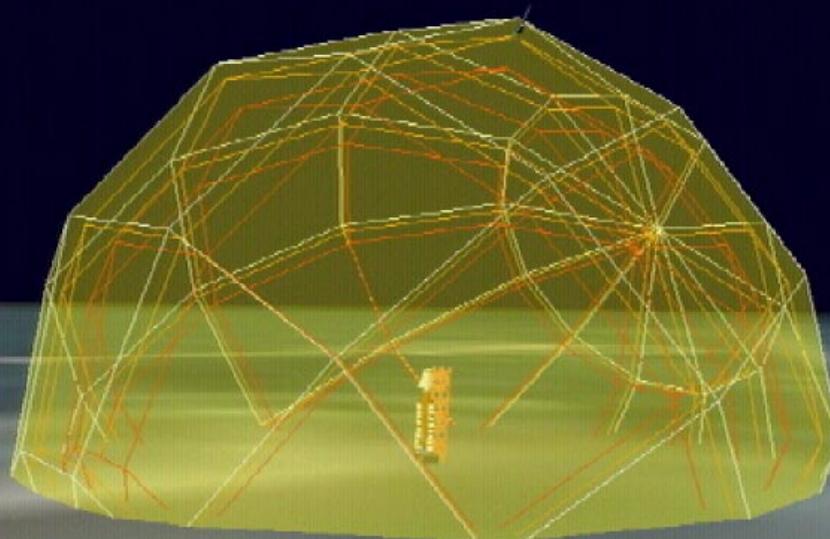


**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

Catastrophic Failure Visualization Sequence

Time: 2.75 sec
SimSpeed: 10.00 %

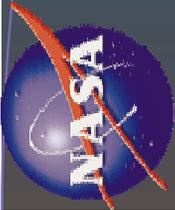
Run ID: 0001
Take-off @ 0.00 sec
Escape activation @ 0.50 sec
Explosion @ 2.45 sec



Blast ID: LH2/LH2 Configuration (Worst Case)
Radius: 668.41 ft
Pressure: 14.05 psi
Temp: 0.00 F

Escape Stage: Escape thruster firing

Contact with CES at - Dist.: 639.25 ft
Press.: 15.18 psi
TOF: 2.23 sec



**JOHNSON
ENGINEERING**
A SPACEHAB COMPANY

CELE CONCEPT EXPLORATION LAB